**RESEARCH**



# **Study of energy-efficient heat resistance and cooling technology for high temperature working face with multiple heat sources in deep mine**

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# **Abstract**

In the present research, we proposed a scheme to address the issues of severe heat damage, high energy consumption, low cooling system efficiency, and wastage of cold capacity in mines. To elucidate the seasonal variations of environmental temperature through feld measurements, we selected a high-temperature working face in a deep mine as our engineering background. To enhance the heat damage control cability of the working face and minimize unnecessary cooling capacity loss, we introduced the multi-dimensional heat hazard prevention and control method called "Heat source barrier and cooling equipment". First, we utilize shotcrete and liquid nitrogen injection to eliminate the heat source and implemented pressure equalization ventilation to disrupt the heat transfer path, thereby creating a heat barrier. Second, we establish divisional prediction models for airfow temperature based on the variation patterns obtained through numerical simulation. Third, we devise the location and dynamic control strategy for the cooling equipment based on the prediction models. The results of feld application show that the heat resistance and cooling linkage method comply with the safety requirement throughout the entire mining cycle while efectively reducing energy consumption. The ambient temperature is maintained below 30 °C, resulting in the energy saving of 10% during the high-temperature period and over 50% during the low-temperature period. These fndings serve as a valuable reference for managing heat damage in high-temperature working faces.

**Keywords** High-temperature working face · Heat source barrier · Multiple heat source efect · Airfow temperature prediction · Dynamic control strategy

# **1 Introduction**

In the last years, industrial and mining technologies have rapidly evolved. Unfortunately, shallow recoverable coal resources have gradually decreased as depth of mining has increased in addition to severe underground thermal pollution (Yao [2018](#page-15-0)). Working faces in deep mines are always characterized by high temperatures of the surrounding rocks, complex and regional concentration of heat sources, and goaf heat intrusion, among others (Liu et al. [2018](#page-14-0); Jayasuriya et al. [2022](#page-14-1)), which result in the emergence of areas with severe underground heat damage. In order to ensure

the safety of workers and production efficiency, heat harm is controlled by applying diferent measures.

At present, heat damage in high-temperature coal faces is prevented by using diferent strategies including heat barrier and mechanical refrigeration. Zou et al. ([2016\)](#page-15-1) used heat insulation shunt cooling to change the heat transfer and airfow structure of roadways, and reduced the airfow temperature on the working face. Chang et al.  $(2021)$  $(2021)$  proposed a cool-wall cooling system, where they placed a heat absorption plate on the roadway wall to absorb and isolate the heat dissipation from the surrounding rock. With this, wind temperature in the roadway was reduced by 3 °C. Liu and Zhang ([2017\)](#page-14-3) developed a bubble composite heat insulation system with expanded perlite as the main material, which reduced the heat dissipation of the roadway surrounding rock. In this case, the temperature of the roadway airfow outlet decreased by 4.956 °C. In addition to blocking the heat dissipation of roadway surrounding rock, the blocking of goaf heat source is also an important mean to control the

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thermal damage of the working face (Wang et al. [2020](#page-15-2)). Xue et al. ([2020\)](#page-15-3) applied a microcapsule foam gel barrier to the goaf, which prevented the spontaneous combustion of the coal left in the goaf and successfully isolated the goaf heat. Akira Yoshida et al. [\(2015](#page-15-4)) determined that inert gasretarded water mist quickly suppressed spontaneous coal combustion. This was successfully applied for fre prevention in coal mine goaf. These heat source barrier methods display good performance; however, the heat barrier efect gradually declines with time. In addition, the barrier cannot meet the cooling demands of ultra-high-temperatures of the working face found in deep levels of mines. In order to solve these problems, many scholars (Zhang et al. [2020;](#page-15-5) Zhai et al. [2019;](#page-15-6) Buys et al. [2015](#page-14-4)) have proposed the use of mechanical refrigeration and other cooling equipment. Brake and Fulker [\(2000\)](#page-14-5), in collaboration with Mount Isa Mines, developed a comprehensive cooling system and supporting solution to deal with the high temperature thermal damage of deep well mining. For this purpose, they measured and calculated different thermal parameters for the optimization of the ventilation and bulk air-cooling systems. Steven Bluhm et al. [\(2014](#page-14-6)) established a complete mine cooling system for Resolution Copper Mine and introduced the main components of the ground and underground cooling systems, such as central surface cooling machines, surface bulk air coolers, and centralized underground cooling machines, among others. The double cooling scheme reduced the threat of heat damage. Stephen G Hardcastle ([2004](#page-14-7)) was able to regulate the underground climate by optimizing the ventilation system. Herein, local cooling measures were implemented according to actual cooling requirements of operations personnel and machines by establishing local microclimate environments to ensure safe operations. Kuyuk et al. [\(2019\)](#page-14-8) designed a deep mine cooling system based on natural lakes, making full use of low temperature lake water for cooling. Guo et al. ([2017\)](#page-14-9) created the HEMS cooling system, which uses underground cold water to absorb heat. With this method, the cooling of high-temperature working faces was performed by taking advantage of geothermal resources. Du Plessis et al. ([2013\)](#page-14-10) proposed a fow-changing energy-saving strategy based on the original mine integrated cooling system, which reduces energy consumption as long as cold is supplied. However, these schemes present several problems including complex system composition, low DOF (degree of freedom) of regulation, and severe cold waste. Because of this, the cooling equipment cannot be moved according to seasonal variations and the cold requirements. Therefore, it is necessary to adopt a new energy-saving and efficient cooling scheme.

According to this analysis and considering the disadvantages of the existing techniques for working face cooling, in the present research we considered both aspects, the heat

source barrier and cooling equipment. First, we propose to use heat source barrier methods for the two main heat sources located in the surrounding rock and goaf of the hightemperature working face. In addition, the layout scheme and intelligent control strategy of the cooling equipment was considered. Finally, we selected the 74104 high-temperature working face of a deep well as engineering background in order to analyze the potential practical application.

### **2 Project overview**

In the present study, the 74104 high-temperature working face of a deep mining well in Jiangsu Province of China was selected as engineering background. The 74104 working face was located at a depth of 1150 m, which corresponds to deep mining conditions. The working face was mined along a strike length of 1374 m and an inclined length of 175 m. The minable coal seam corresponded to the No. 7 coal seam, with an average thickness of 3.18 m, and average dip angle of 23°.The coal seam where the 74104 working face was located had a spontaneous ignition grade of Class I, which indicated that coal seam was prone to spontaneous combustion. The mining of the 74104 working face began on Jun 1st, 2020, with a daily advance speed of 3 m/d and air volume ventilation of  $1400 \text{ m}^3/\text{min}$ . The original rock temperature reached 42 °C, indicating a second-degree heat hazard level.

In order to refect the change rule of airfow temperature more directly and formulate an optimal cooling scheme for high-temperature working faces, temperature and humidity sensors were used for feld measurement. These measurements were performed in seven diferent points of the hydraulic supports located in the working face (nearly 10 m one from the other) and at intervals of 200 m in headentry and tailentry. In addition, measurements were performed at the inlet and outlet of airfow in each region. Six representative measurement points on the ventilation route of the 74104 working face were calibrated. Point 1 was located in front of the cooling fan in the subinclined shaft; point 2 was located at the inlet of the 74104 headentry; point 3 was located at the 500 m milestone of the 74104 headentry; point 4 was located at the inlet of the 74104 working face; point 5 was located at the center of the 74104 working face; and point 6 was located at the outlet of the 74104 working face. Measurement points are shown in Fig. [1.](#page-2-0)

Mining of the 74104 working face began on June 1st, 2020. In addition, the airfow temperature measurements were performed every seven days starting June 2nd. Temperatures were taken between 12:00 p.m. and 14:00 p.m., where the highest temperatures occur.



<span id="page-2-0"></span>**Fig. 1** Location of the points where meteorological parameters where measured

# **3 Heat barrier method**

The frst step considered the partial obstruction of the heat source in the high-temperature working face, which reduced both, heat dissipation and refrigerant pressure of the cooling equipment. The heat barrier scheme considered the heat dissipation of surrounding rock and goaf. For this purpose, shotcrete and water spray were used to reduce the heat dissipation of surrounding rock. In addition, a wall was used to obstruct the heat dissipation of transported coal and gangue (Wang and Zhou [2020](#page-15-7)). In order to eliminate heat source, spontaneous combustion of residual coal was hindered (Zhang et al.  $2019$ ). Also, in order to cut off heat transfer, air leakage was reduced.

Prior to the working face mining, the sampling beam tubes were placed at diferent intervals in the tailentry, these sampling beam tubes successively entered the goaf with the advance of the working face (Ray and Singh [2007\)](#page-14-11). Later, the location of the "spontaneous combustion zone" can be identifed by extracting and analyzing the gas in the beam tube to obtain the oxygen concentration in diferent areas. Finally, liquid nitrogen was injected into the "spontaneous combustion zone" and goaf heat was eliminated from the source.

The pressure equalization and ventilation technologies are used to decrease the rate of air leakage during the mining of the working face. However, this method results in the increase of air leakage into the goaf and spontaneous combustion of residual coal. In order to solve the contradiction between two key points, it is necessary to determine the optimal pressure difference  $\Delta P$  and the position of pressure equalization point.

Because of the even-distributed characteristics of wind resistance, the air pressure decreases linearly in the working face (Tamminen et al. [2016\)](#page-15-9). First, assuming the air pressure in the inlet side is  $P_a$ , the air pressure in the return side (upper corner) is  $P_b$ , and the air pressure inside the goaf maintains a stable value  $P_g$ . Then, the pressure equalization point is  $x_0$ , and  $P_0$  equals to  $P_g$  at this point. In order to reach the optimal pressure diference Δ*P* and the position of the pressure equalization point, the upper corner of the working face is taken as the coordinate origin to establish a rectangular coordinate system. As seen in Fig. [2,](#page-3-0) the x-axis represents the length of any point from the upper corner, and the *y*-axis represents the amount of air leakage at any point. Moreover,  $P_g$ ,  $P_a$ , and *L* values are known.

$$
P_0 = P_a - \frac{\Delta P}{L}(L - x_0) = P_g \tag{1}
$$

<span id="page-2-1"></span>
$$
\frac{\Delta P}{L} = \frac{P_a - P_g}{L - x_0} \tag{2}
$$



<span id="page-3-0"></span>**Fig. 2** Relationship between air leakage rate and relative position under pressure equalization ventilation

The diferential pressure gap method was used to calculate the air leakage (Lolon et al. [2017;](#page-14-12) Wu et al. [2019](#page-15-10)).

$$
S = 0.67 \times A \times \sqrt{\Delta P} \tag{3}
$$

Here, *S* is the air leakage, *A* is the air leakage area, and Δ*P* is the diferential pressure. Later, the Equation used to calculate the total air leakage caused by the pressure diference between goaf and working face can be obtained:

$$
S = \int_0^{x_0} 0.67A \sqrt{P_g - \left[ P_a - \frac{\Delta P}{L} (L - x) \right]} dx
$$
  
+ 
$$
\int_{x_0}^L 0.67A \sqrt{P_a - \frac{\Delta P}{L} (L - x) - P_g} dx
$$
 (4)

Equation [\(5](#page-3-1)) is obtained by incorporating Eq. ([2](#page-2-1)) into Eq. ([4\)](#page-3-2) and making  $m = \frac{P_a - P_g}{L - x_0}$ .

$$
S = 0.67A \int_0^{x_0} \sqrt{mx_0 - mx} dx + 0.67A \int_{x_0}^L \sqrt{mx - mx_0} dx
$$
\n(5)

Solving the defnite integral of the right-hand side of the equation and making  $\sqrt{mx_0 - mx} = z$ ,  $\sqrt{mx - mx_0} = t$ .

$$
S = 0.67A \int_{\sqrt{mx_0}}^0 -\frac{2z^2}{m} dz + 0.67A \int_0^{\sqrt{mL - mx_0}} \frac{2t^2}{m} dt
$$
 (6)



<span id="page-3-5"></span>**Fig. 3** Function  $G(x_0)$ 

<span id="page-3-3"></span>
$$
S = 0.67A \left[ \frac{2(\sqrt{mL - mx_0})^3}{3m} + \frac{2(\sqrt{mx_0})^3}{3m} \right]
$$
(7)

When  $m$  is introduced into Eq.  $(7)$  $(7)$  $(7)$ , the final Eq.  $(8)$  $(8)$  $(8)$  can be obtained.

<span id="page-3-4"></span>
$$
S = 0.447A \times (P_{\rm a} - P_{\rm g})^{\frac{1}{2}} \left[ (L - x_0) + \frac{x_0^{\frac{3}{2}}}{(L - x_0)^{\frac{1}{2}}} \right]
$$
(8)

$$
G(x_0) = (L - x_0) + \frac{x_0^{\frac{3}{2}}}{(L - x_0)^{\frac{1}{2}}}
$$
\n(9)

Equation ([8](#page-3-4)) can be decomposed in two parts, the frst half is a constant and the second half is a function of  $x_0$ . Figure [3](#page-3-5) represents the function  $G(x_0)$  when *L* equals 175 m.

<span id="page-3-2"></span>The minimum total air leakage of the working face can be obtained through derivation of the function  $G(x_0)$  without considering actual engineering requirements.

$$
G'(x_0) = 1.5x_0^{\frac{1}{2}}(175 - x_0)^{-\frac{1}{2}} + 0.5x_0^{\frac{3}{2}}(175 - x_0)^{-\frac{3}{2}} - 1 \quad (10)
$$

<span id="page-3-1"></span>The function  $G(x_0)$  decreases monotonically when  $x_0$  ranges from 0 to 45.657 m, but increases monotonically when  $x_0$  ranges from 45.657 to 175 m. Therefore, the air leakage of the working face reaches the minimum 69.9475*A*  $\times (P_a - P_g)^{\frac{1}{2}}$  as the pressure equalization point is located 45.657 m from the upper corner. The pressure difference  $\Delta P$  between the two ends of the working face is 1.353( $P_a$ - $P_g$ ). In addition, the upper corner should be set as the pressure equalization point in order to improve the heat barrier effect. On the other hand, to ensure the best

heat barrier effect while reducing spontaneous combustion of residual coal, the pressure equalization point should be moved away from the upper corner. Selection of the specifc location of the pressure equalization point should be based on multiple factors including the advancing speed of working face, the speed of air leakage, spontaneous combustion period of coal seam, and gas emissions.

Considering this analysis, the pressure equalization ventilation system in high-temperature working face was designed and the specifc layout is shown in Fig. [4](#page-4-0).

First, it is necessary to determine the periodic variation of air pressure in the goaf or relatively stable  $P_{\alpha}$  value. Next, the local fan model and reasonable working condition can be confrmed according to the ventilation volume required for safe production. The initial air pressure of the local fan is  $P_f$ ; the air pressure at the air outlet duct is represented by  $P_1$ ; the air pressure at the inlet of the working face  $P_a$  can be obtained with consideration of the frictional resistance. Third, the air pressure at the outlet of the working face  $P_b$  can be calculated using pressure diferences Δ*P* and the Equations previously shown. In addition, with friction resistance and air pressure variations, proper adjustment of the local fan and air window ensure an optimal pressure equalization ventilation mode. Taking 74104 high-temperature working face as an example, the optimized pressure equalization ventilation mode can efectively block the intrusion of goaf heat. Our results indicated that air temperature was reduced by 0.5–0.7 °C after the application of the optimized pressure equalization ventilation (Fig. [5](#page-4-1))**.**



<span id="page-4-0"></span>**Fig. 4** Specifc layout of pressure equalization ventilation system in high-temperature working face



<span id="page-4-1"></span>**Fig. 5** Airfow temperature before and after the implementation of pressure equalization ventilation

# **4 Numerical simulation**

# **4.1 Model setup**

The working face is complex and contains multiple heat sources. Because of the influence of different heat sources such as high-temperature ribs and mechanical equipment, the temperature of the fresh airflow gradually increased. In this case, temperature measurements are inaccurate since the installation of the temperature and humidity sensor is difficult. Since the second step considers the design of the cooling system, it is important to determine the effect of multiple heat sources on airflow temperature. With this information, a more accurate system can be designed.

Figure [6](#page-5-0) shows the airflow heat exchange model of the working face built using the COMSOL software. The model consisted of several cuboids with a length of 175 m, a width of 5 m, and a height of 4 m. The parameters were set up according to the distribution of the heat source in the working face. Thermal contribution of the personnel, coal cutter, and other heat sources all belong to absolute heat sources and are located in the same position. Thus, in the simulation, they can be considered as an integral heat source (Tu et al. [2021](#page-15-11)). The heat dissipation of coal and pillar rib transportation are averaged. In this case, temperature of coal transportation was 37 °C, the rib temperature 42  $\degree$ C, and the convective heat transfer coefficient 6.565  $W/(K \text{ m}^2)$ . The upper, lower, and left surfaces of the model were considered adiabatic walls. The heat dissipation of the goaf can be ignored after heat source barrier measures were implemented.

The temperature of the inlet airflow, air velocity, and rib temperature have an impact on the change of airfow temperature through the analysis of the heat exchange process of the working face. In order to analyze these efects, the control variate method was selected to set diferent initial parameters for the numerical simulation as illustrated in Table [1](#page-5-1).

#### **4.2 Variation of airfow temperature**

Data in Fig. [7](#page-6-0)a indicate that the variation of airflow temperature in the working face can be divided into three stages. Between 0 and 90 m, the airflow temperature increases linearly with the increase in fow distance. Later, the increase trend changed between 90 and 110 m. This occurred because there is an absolute heat source concentration area where the

<span id="page-5-1"></span>**Table 1** Infuencing factors of airfow temperature in working face

Group	Inlet airflow tempera- ture $(^{\circ}C)$	Air velocity (m/s)	Rib tem- perature $(^{\circ}C)$
1	21 24	1.5	42
	28		
2	24	1.5 2.0 2.5	42
3	24	1.5	40 42 44



<span id="page-5-0"></span>**Fig. 6** Model of heat transfer between the airfow and the working face



<span id="page-6-0"></span>**Fig. 7 a** Unitive transformation law **b** Under diferent air inlet temperatures **c** Under diferent rib temperatures **d** Under diferent air velocities

heat increases signifcantly. Between 110 and 175 m, the airfow temperature still increases with the increase in fow distance; however, the increase amplitude diminished gradually until it approached to zero. At the fnal stage, the airfow temperature was close to 30 °C. When the temperature of the inlet airflow was  $21 \degree C$ , that at the back of the working face slightly exceed 30 °C. This means that some cooling measures should be taken to ensure the safety of the mine as the inlet airfow temperature continues to rise.

Results in Fig. [7](#page-6-0)b–d indicate that the inlet airfow temperature, air velocity, and rib temperature afected airfow temperature. These data also showed that the higher the inlet airfow temperature, the higher the temperature of the returning airfow, and the smaller the amount of heat-exchanged between the airfow and the environment (Fig. [7](#page-6-0)b). Therefore, this method can be used to accurately monitor the temperature of the inlet airflow and determine if the working face requires to be cooled down. The results presented herein provide a model to estimate the temperature of the airfow and an intelligent strategy to control cooling devices.

Figure [7c](#page-6-0) indicates that, within the same flow distance, the higher the rib temperature, the larger the increase in airfow temperature, and the greater the value of the airfow temperature when stability is reached. In addition, the overall temperature of the airfow will not change in spite of variations in rib temperature. Moreover, a smaller diference in airfow temperatures were observed when rib temperatures were 44 °C and 42 °C, as compared to those of 42 °C and 40 °C. These data indicated that rib temperature was negatively correlated with Δ*t* and the airfow temperature did not infnitely increased as rib temperatures also increased *(*Δ*t is the diference of airfow temperatures corresponding to diferent rib temperatures)*.

Figure [7d](#page-6-0) indicates that the larger the air velocity, the smaller the increase in airfow temperature within the same flow distance, and the smaller the stable airflow temperature fnally reached. This means that the environmental temperature of the working face can be reduced by increasing the amount of air. Our data indicate that the best way to cool down the high-temperature working face is by using a cooling equipment adapted with a supporting fan.

# **5 Results and discussion**

#### **5.1 Model of airfow temperature**

In order to achieve a reasonable design of the cooling system and intelligent control of the cooling equipment, the accurate prediction of airfow temperature is required. Thus, in the present research, we obtained the airfow temperature prediction models of headentry, tailentry, and working face in diferent regions. We selected the airfow temperature prediction models of working face as example for detailed analysis.

The diferential equation of heat balance in the working face was based on the law of conservation of energy. The amount of heat exchange between the heat source and airflow in the micro element is equal to the sensible heat of the

airfow (manifested as the increase in airfow temperature) plus the latent heat of water vaporization (Qin and Xu [1998](#page-14-13)).

<span id="page-7-2"></span>
$$
dQ = M_B di = M_B c_p \Delta t + M_B \gamma dx \qquad (11)
$$

where, d*Q* indicates the amount of heat exchange between the heat source and airfow in micro element of the working face;  $M_B$  is the air mass flow rate;  $c_p$  corresponds to the specifc heat at constant pressure; *γ* is the latent heat of vaporization; *x* indicates moisture content; *i* is the enthalpy of incoming airflow; and  $\Delta t$  is the increment of airflow.

Considering the heat sources in the working face, the heat transferred by the thermal environment to the airfow contained in the micro-element of the working face is determined using Eq. ([12\)](#page-7-0):

<span id="page-7-0"></span>
$$
dQ = K_{\tau} U(t_{w} - t) dy + \frac{Q_{s} + Q_{c} + Q_{h} + Q_{r} + Q_{z}}{L} dy
$$
 (12)

Rib and coal transportation are both heat sources, and the heat transferred from coal transportation to airfow is calculated using Eq. ([13\)](#page-7-1) (Zhu et al. [2020\)](#page-15-12):

<span id="page-7-1"></span>
$$
Q_{\rm s} = G_{\rm s} c_{\rm s} \Delta t = 0.0024 L^{0.8} G_{\rm s} c_{\rm s} (t_{\rm avg} - t_{\rm fh}) \tag{13}
$$

There is an approximate linear relationship between moisture content of airfow and airfow temperature at standard atmospheric pressure (Danko et al. [2020\)](#page-14-14).

<span id="page-7-3"></span>
$$
x = \left(\varphi_1 + \frac{\Delta \varphi}{L}y\right)At + \left(\varphi_1 + \frac{\Delta \varphi}{L}y\right)D
$$
  
\n
$$
\Rightarrow dx = \left(\varphi_1 + \frac{\Delta \varphi}{L}y\right)Adt + \frac{\Delta \varphi}{L}(At + D)dy
$$
\n(14)

Solving the simultaneous Eqs.  $(11)$  $(11)$ ,  $(12)$  $(12)$ ,  $(13)$  $(13)$ ,  $(14)$  and after the simplifcation process, we obtained Eq. [\(15](#page-7-4)):

<span id="page-7-4"></span>
$$
\begin{cases}\nE = 1 + \frac{A\gamma}{c_{\text{p}}} \varphi_2 \\
F = \frac{A\gamma \Delta \varphi}{c_{\text{p}} L} \\
\eta = 0.0024 L^{0.8} G_s c_s \\
\Sigma Q_{\text{M}} = Q_{\text{c}} + Q_{\text{h}} + Q_{\text{r}} + Q_{\text{z}}\n\end{cases} (15)
$$

The nonhomogeneous first order linear differential equation with only t and y as the variable parameters was obtained as shown in Eq. (16):

$$
\frac{\mathrm{d}t}{\mathrm{d}y} + \frac{\left(\frac{\eta\sigma}{LM_{\mathrm{B}}c_{\mathrm{p}}} + \frac{K_{\mathrm{r}}U}{M_{\mathrm{B}}c_{\mathrm{p}}} + \frac{\gamma\Delta\varphi A}{Lc_{\mathrm{p}}}\right)}{E + Fy}t = \frac{\frac{K_{\mathrm{r}}Ut_{\mathrm{w}}}{M_{\mathrm{B}}c_{\mathrm{p}}} + \frac{\eta t_{\mathrm{avg}}}{LM_{\mathrm{B}}c_{\mathrm{p}}} + \frac{\Sigma Q_{\mathrm{M}}}{LM_{\mathrm{B}}c_{\mathrm{p}}} - \frac{\Delta\varphi\gamma D}{Lc_{\mathrm{p}}}}{(16)}
$$

The relationship between the working face airfow temperature t and the working face airfow distance y is obtained by solving the previous diferential equation, as seen in Eq. ([17\)](#page-8-0).

$$
t = \frac{K_{\tau}Ut_{w}L + \eta t_{avg} + \Sigma Q_{M} - M_{B}\Delta\varphi\gamma D}{K_{\tau}UL + \eta\sigma + M_{B}\Delta\varphi\gamma A} + \left[ \left( t1 - \frac{K_{\tau}Ut_{w}L + \eta t_{avg} + \Sigma Q_{M} - M_{B}\Delta\varphi\gamma D}{K_{\tau}UL + \eta\sigma + M_{B}\Delta\varphi\gamma A} \right) \times E^{\frac{K_{\tau}UL + \eta\sigma + M_{B}\Delta\varphi\gamma A}{M_{B}\Delta\varphi\gamma A}} \right] \times (E + F y)^{-\frac{K_{\tau}UL + \eta\sigma + M_{B}\Delta\varphi\gamma A}{M_{B}\Delta\varphi\gamma A}}
$$
\n(17)

where,  $t_w$  is the rib temperature;  $t_{avg}$  is the average temperature during coal transportation;  $t_{\text{fh}}$  indicates the wet bulb temperature of airflow,  $t_{\text{fh}} = \sigma t$ ,  $\sigma$  equals to 0.904;  $t_1$  indicates the airflow temperature in working face inlet;  $G<sub>s</sub>$  is the coal mass flow rate;  $c_s$  corresponds to the specific heat of coal;  $K_{\tau}$  is the unstable heat transfer coefficient; *U* represents the perimeter of the working face section; *L* is the length of the working face;  $\varphi_1$  is the relative humidity of inlet airflow;  $\Delta \varphi$ is the diference between the relative humidity of airfow at the inlet and that at the outlet; and  $\Sigma Q_M$  is the sum of the absolute heat exothermic quantity in the working face. The airfow temperature at each point of the working face can be solved by using Eq.  $(17)$  $(17)$ .

The airfow temperature prediction model of headentry can be obtained in a similar way, as seen in Eq. ([18\)](#page-8-1). Herein,  $t_0$  represents the airflow temperature in headentry inlet and  $\Sigma Q_{\text{M1}}$  represents the sum of the absolute heat exothermic quantity in the headentry.

$$
t = \frac{K_{\tau1} U_1 t_{w1} L_1 + \Sigma Q_{M1} - M_B \Delta \varphi \gamma D}{K_{\tau1} U_1 L_1 + M_B \Delta \varphi \gamma A} +
$$
  
\n
$$
\left[ \left( t_0 - \frac{K_{\tau1} U_1 t_{w1} L_1 + \Sigma Q_{M1} - M_B \Delta \varphi \gamma D}{K_{\tau1} U_1 L_1 + M_B \Delta \varphi \gamma A} \right) \times E^{\frac{K_{\tau1} U_1 L_1 + M_B \Delta \varphi \gamma A}{M_B \Delta \varphi \gamma A}} \right] \times (E + F y)^{-\frac{K_{\tau1} U_1 L_1 + M_B \Delta \varphi \gamma A}{M_B \Delta \varphi \gamma A}}
$$
\n(18)

The model to predict airflow temperature at the tailentry is presented in Eq.  $(19)$  $(19)$ . In this case,  $t_2$  represents the airflow temperature at inlet tailentry and  $\Sigma Q_{\text{M2}}$  represents the sum of the absolute heat of the exothermic process at the tailentry.

$$
t = \frac{\eta t_{\text{avg}} + \Sigma Q_{\text{M2}} - M_{\text{B}} \Delta \varphi \gamma D}{\eta \sigma + M_{\text{B}} \Delta \varphi \gamma A} + \left[ \left( t_{2} - \frac{\eta t_{\text{avg}} + \Sigma Q_{\text{M2}} - M_{\text{B}} \Delta \varphi \gamma D}{\eta \sigma + M_{\text{B}} \Delta \varphi \gamma A} \right) \times E^{\frac{\eta \sigma + M_{\text{B}} \Delta \varphi \gamma A}{M_{\text{B}} \Delta \varphi \gamma A}} \right] \times (E + Fy)^{-\frac{\eta \sigma + M_{\text{B}} \Delta \varphi \gamma A}{M_{\text{B}} \Delta \varphi \gamma A}}
$$
(19)

<span id="page-8-0"></span>

<span id="page-8-3"></span>**Fig. 8** Airfow temperatures at diferent measuring points in diferent periods

## **5.2 Accuracy of prediction model**

In order to ensure that the model to predict airfow temperature can be applied to the whole production cycle of the mine and that it is able to resemble underground temperature changes caused by seasonal variations as well as thermal flywheel effect and others, we analyzed the measured data and identifed the dominant factors afecting underground temperature changes. Based on this, the measured data of airfow temperature with certain seasonal representativeness were selected for model verifcation. The measured wind temperature at each measuring point in diferent periods of 74104 working face are shown in Fig. [8.](#page-8-3)

<span id="page-8-1"></span>Figure  $8$  shows that the seasonal variation will have a signifcant impact on temperature levels in underground environments. During the summer, the underground also faces high temperatures and heat damages, especially from June to August. Even after cooling, the wind temperature of the 74104 working face will still exceed 30 °C, which requires secondary cooling measures. In autumn, from September to mid-November, high temperature periods lead to thermal damage events; however, the overall climate of the mine is good. In winter, from late November to mid and late February, safe mining can be ensured without refrigeration and cooling measures. Spring runs from March to May. In this case, the situation is similar to autumn. Therefore, selected data should include that corresponding to summer, autumn (or spring), and winter.

<span id="page-8-2"></span>As shown in Fig. [8,](#page-8-3) the critical value of the ground temperature is 30 °C. When temperature is higher than 30 °C, the airfow temperature gradually decreases in the process of underground flow. On the contrary, the airflow temperature increases gradually, which is caused by the thermal fywheel efect (Scalise et al. [2021](#page-14-15); Roghanchi and Kocsis [2019\)](#page-14-16). This efect has a certain infuence on the arrangement of ventilation and cooling system. Therefore, it is necessary to consider this point when selecting reasonable measured wind temperature data. In addition, the ground weather changes and diurnal changes will also have a certain impact on the selection of measured data and the verifcation of the prediction model. Nevertheless, the changes can be included in the temperature range of the four seasons, and not considered as an independent factor in the selection of measured data.

Based on this, we divided the whole production cycle of working face into four representative periods: above 30 °C (mainly in summer), 20–30 °C (summer and part of spring and autumn),  $10-20$  °C (mainly in spring and autumn), and below 10 °C (mainly in winter). Therefore, we selected the measured air fow temperature data on June 22nd (summer), August 3rd (summer), October 19th (autumn) and November 30th (winter) to verify the wind temperature prediction model (Tables [2](#page-9-0) and [3\)](#page-9-1). In addition, June 2nd was selected as the starting time of data recording, and this date belongs to the period above 30 °C. Wind temperature recorded on June 2nd was considered the preliminary reference basis for the cooling and cooling system layout of the 74104 working face, which provide safe cooling conditions of 74104 working face. The optimal cooling system layout scheme needs to further consider the cooling demand in the ultra-high temperature period in summer. The cooling system layout method will be analyzed and introduced in detail in Section 4.4.

Multiple groups of calculated data and measured data were plotted using Origin. The results are shown in Fig. [9](#page-9-2).

The results indicated that the model to predict airfow temperature can be used with a certain margin of error. The cooling capacity of the working face can be obtained through the temperature airfow prediction results, which provides a theoretical basis for the formulation of intelligent control strategy of cooling equipment.

<span id="page-9-0"></span>**Table 2** Airfow temperature in the intake roadway of 74104 working face

Flow distance (m)	June 22nd			August 3rd October 19th November 30th
$\mathbf{0}$	26.0	28.2	18.5	18.0
200	26.4	28.5	19.1	18.5
400	26.7	28.6	19.6	19.0
500	26.8	28.5	19.8	19.2
600	26.9	28.7	20.0	19.5
800	27.2	28.8	20.5	19.9
1000	27.4	29.0		
1200	27.7			
1300	27.8			

<span id="page-9-1"></span>**Table 3** Airfow temperature of the 74104 working face

Flow distance (m)	June 22nd			August 3rd October 19th November 30th
$\mathbf{0}$	27.7	29.2	20.6	20.1
20	28.1	29.6	21.7	21.2
40	28.6	30.2	22.5	22.0
60	29.3	30.7	23.5	23.2
80	29.6	31.2	24.4	24.0
100	30.1	31.7	25.3	24.8
120	30.6	32.3	26.0	25.7
140	31.0	32.6	26.7	26.5
160	31.4	32.7	27.4	27.5
175	31.4	32.8	28.0	27.8



<span id="page-9-2"></span>**Fig. 9 a** Comparison between the measured and calculated airfow temperature in headentry **b** Comparison between the measured and calculated airfow temperature in the working face

## **5.3 Model applicability analysis**

After verifying the accuracy of the airfow temperature prediction model, it is necessary to analyze its applicability and determine its potential limitations. The model was determined considering ventilation parameters, roadway layout parameters, and thermal physical parameters of the heat source. Therefore, when the roadway layout or ventilation mode of the working face changes, the matching airfow temperature prediction model can be established by adjusting relevant parameters. Unfortunately, the current model is more applicable to underground mining because it was built considering the ventilation of shafts and roadways. Since the previous research and data measurement were performed for coal mining conditions, it is impossible to clarify the applicability of the model and methods for non-coal mining cases. However, further research will be performed in order to provide a reference value for the establishment of the airfow temperature model of non-coal mining working faces in underground mining.

#### **5.4 Industrial application**

In the present research, an energy-saving and efficient cooling scheme for high-temperature working faces in deep mines was proposed. For this purpose, the previous theoretical analysis and feld measurements were considered. The system included the selection of the initial position of the cooling equipment and the intelligent control strategy by considering working face distance and seasonal airfow temperature variations. Field test of cooling scheme was carried out relying on the actual engineering background of 74104 high-temperature working face.

Zhangshuanglou Coal Mine can be divided into the east wing and the west wing mining area. Each wing is equipped with a set of German Pegasus KM2000 underground centralized chiller and a KM1000 unit in the−750 m main refrigeration chamber, with a total refrigeration capacity of 3000 kW. The 74104 working face is located in the west wing. According to the mining and excavation replacement plan of the mine, the cooling system of the west wing mine needs to conduct refrigeration and cooling operations on the 74104 working face, 94102 heading face and 94602 heading face at the same time. Therefore, the total cooling capacity that can supply the 74104 working face is 1000 kW. The working face started mining activities on June 1st, and the air fow temperature at the inlet of the inlet tunnel of the working face was 29 °C on June 2nd. To ensure that working face production is carried out in a safe environment, the relationship between the cooling capacity required for the working face and the layout position were determined. Results are shown in Table [4](#page-10-0).

According to data presented in Table [4](#page-10-0) (characters in bold), when the cooling equipment was placed 1000 m away from the airfow inlet of the working face, the cooling capacity was

<span id="page-10-0"></span>

970 kW, which is smaller than the 1000 kW required for cooling the working face. In addition, the working face is not able to reach the cooling demands as the distance increases. Therefore, the cooling equipment is initially placed in the refrigeration chamber within 1000 m from the inlet of the working face. Also, the cooling equipment layout needs to consider whether it can comply with the cooling demand of the working face during the high temperature period. The maximum airfow temperature was found from the recorded data for further analysis. The measured airfow before cooling was 30.6 °C (The hottest time of the year) and the advance distance was 210 m on August 10th. Then, the airfow temperature after cooling can be calculated according to the cooling capacity of 1000 kW, and fnally the change of return airfow temperature of working face under diferent layout positions are obtained, as seen in Table [5.](#page-11-0)

Table [5](#page-11-0) (Bold lines) indicates that the returning airflow temperature is almost close to the critical temperature of 30 °C when the cooling equipment is 590 m away from the inlet of the working face. In conclusion, with the working face advance distance of 210 m, the cooling equipment ought to be arranged in the refrigeration chamber within 800 m from the inlet of the working face at the beginning of mining activities. This set up will provide the proper cooling conditions during the whole mining cycle, without the need of frequently "moving" the equipment. The optimal efect can be achieved with this scheme.

In order to achieve safe conditions, energy conservation, and high efficiency, it is necessary to formulate an intelligent strategy for controlling the cooling equipment by considering the initial position of cooling equipment. The cooling capacity is dynamically adjusted with seasonal variations and advancement of the working face. Table [6](#page-11-1) presents the return airfow temperature under diferent cooling capacities in late September. The average airfow temperature before cooling is 25 °C and the advance distance is 345 m at this moment.

<span id="page-11-0"></span>**Table 5** Airfow temperature of headentry and tailentry under diferent layouts



#### <span id="page-11-1"></span>**Table 6** Return airflow temperature under diferent cooling capacities





<span id="page-11-2"></span>**Fig. 10** Intelligent control strategy of cooling equipment





<span id="page-12-0"></span>**Fig. 11** Software operation interface

Table [6](#page-11-1) (Bold lines) indicates the cooling capacity of 265 kW can meet the cooling demand of the working face in late September, the return airfow temperature is lower than the critical temperature of 30 ℃ under this condition. The airflow temperature at inlet of headentry and the flow distance decreased gradually with seasonal variations and advancement of the working face. As Table [6](#page-11-1) shows, safety mining can be performed without opening the cooling



<span id="page-13-0"></span>**Fig. 12 a** Airfow temperature before and after the implementation **b** Cold quantity demand and energy-saving efciency in the full mining cycle

equipment when the airflow temperature is less than 22 °C. Thus, the intelligent control strategy of the cooling equipment can be summarized. "A period of ten days" was taken as a time to adjust the cooling capacity and appropriately reduce the period of regulation in the active phase of airfow temperature variation. Besides, an intelligent control software with intelligent control strategy was developed to facilitate feld application, as seen in Figs. [10](#page-11-2) and [11](#page-12-0).

The airfow thermal parameter of the working face was measured after the cooling scheme and intelligent control strategy were applied to the 74104 working face. Figure [12](#page-13-0)a presents the change of airfow temperature before and after cooling. In addition, the average cooling capacity was determined on a period of ten days rotation, and the saving energy consumption of refrigeration after the application of the new cooling scheme was calculated, as seen in Fig. [12](#page-13-0)b.

Figure [12](#page-13-0)a indicates that the thermal hazard threat has been signifcantly reduced, the return airfow temperature can be maintained at 29.4 °C, which is 4 °C lower than that before the implementation of cooling scheme. Figure [12b](#page-13-0) indicates that the safety mining can be guaranteed without any cooling measures after November. In addition, a 10% in energy saving efficiency can be reached during the high temperature period, and more than 50% in low temperature period after the application of the new intelligent cooling scheme. The new cooling scheme can dramatically reduce the waste of power energy. This complies with the requirements for safe mining, high efficiency, and green and energy conservation.

# **6 Conclusions**

In the present research, we selected the 74104 high-temperature working face as engineering background. After combining theoretical analysis, numerical simulation, and feld investigation, we obtained the following conclusions.

- (1) Given the characteristics of complex heat sources, scattered heat source locations, and severe regional heat damage threats in high-temperature working faces, we proposed a combined scheme for heat hazard prevention and control described as "heat source barrier + cooling equipment". In addition, the present research focused on the heat barrier method of pressure equalization ventilation and dynamic control strategy of the cooling equipment.
- (2) The numerical simulation method was used to obtain the variation rule of airfow temperature afected by multiple heat sources. We proposed models to predict airfow temperature of headentry, tailentry, and working face by considering the variation rule of airfow temperature. Then, and automatic control software was designed for the 74104 high-temperature working face. The cooling equipment was arranged in the refrigeration chamber within 800 m from the inlet of the working face and the period of regulation was "a period of ten days".
- (3) After the application of the proposed composite cooling scheme, the ambient temperature was always below 30 °C, and the energy saving efficiency reached  $10\%$ during the high-temperature period and more than 50% in the low temperature period. The scheme proposed herein properly complies with the requirements of safety, high efficiency, and green and energy conservation. In addition, it provides reference for similar heat harm control in high-temperature working faces.

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# **Declarations**

**Competing interests** The authors declare that they have no competing interests.

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